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SECTION II

LIMITATIONS

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AIRPLANE FLIGHT MANUAL
R90 - 230 RG

Ruschmeyer

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SECTION II

LIMITATIONS

INTRODUCTION

This section of the Airplane Flight Manual presents the FAA approved operating limitations, the significance of such limitations, instrument markings and basic placards necessary for the safe operation of the airplane, its powerplant, standard systems and standard equipment. Optional equipment may require additional operating limitations and placards, which are presented in Section IX of this manual.

CERTIFICATION BASIS

The RUSCHMEYER R 90-230 RG is certificated in the "Normal" Category according to FAR Part 23, including Amendments 23-1 to 23-34.

AIRSPEED LIMITATIONS

Airspeed	KCAS	KIAS	Remarks
Maneuvering Speed V _A	128* (2977 lbs/ 1350 kg) 124 (2803 lbs/ 1271 kg) 108 (2119 lbs/ 961 kg)	128 124 107	Do not make abrupt control movements above this speed.
Maximum Flap Extended Speed V _{FE}	103	102	Do not exceed this speed with the given flap setting.
Maximum Landing Gear Operating Speed VLO	14()	140	Do not extend or retract landing gear above this speed.
Maximum Landing Gear Extended Speed VLE	140	140	Do not exceed this speed with landing gear extended.
Never Exceed Speed V _{NE}	194* (MSL - 12.000 ft) 182 (16.000 ft) 171 (20.000 ft)	193 181 171	Do not exceed this speed in any operation at any time.
Maximum Structural Cruising Speed V _{NO}	157	157	Do not exceed this speed except in smooth air and then only with caution.
Maximum Airspeed, Cowl Flaps Open	194	193	Do not exceed this speed with cowl flaps open.

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AIRSPEED INDICATOR MARKINGS

Marking	KIAS	Significance
White Arc	60 - 102	Operating range with wing flaps extended. Lower limit is maximum weight stalling speed in landing configuration. Upper limit is maximum speed permissible with flaps extended.
Green Arc	71 - 157	Normal operating range. Lower limit is maximum weight stalling speed with flaps and landing gear retracted. Upper limit is maximum structural cruising speed.
Yellow Arc	157 - 193	Caution range. Operations must be conducted with caution and only in smooth air.
Red Radial	193	Maximum speed for all operations.

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POWERPLANT LIMITATIONS

ENGINE: Lycoming IO-540-C4D5

Engine Operating Limits for Take-Off and Continuous operation:

	•	
Maximum Power		172 kW (231 HP)
Max. Permissibl	e Propeller Speed	2 400 RPM
Maximum Manifold Pressure @ prop. speed 1800 RPM 1900 RPM 2000 RPM 2100 RPM 2200 - 2400 RPM		25 in Hg* 26 in Hg* 27 in Hg* 28 in Hg* maximum
Maximum Cylin	der Head Temperature	260 °C (500 °F)
Maximum Oil Temperature		118 °C (245 °F)
Minimum Oil Pressure		1.7 bar (25 psi)
Maximum Oil Pressure		7.9 bar (115 psi)
Minimum Fuel Inlet Pressure at Fuel Injection System Normal Operation Idle		0.96 bar (14 psi) 0.83 bar (12 psi)
Maximum Fuel Inlet Pressure at Fuel Injection System		3.1 bar (45 psi)
Maximum Fuel Nozzle Pressure		0.56 bar (8.0 psi)

^{*} Linear Interpolation between given values

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PROPELLER:

Mühlbauer MTV-14-B/190-17

Diameter:

74.8 in $(1.90 \text{ m})^{+0.0 \text{ in } (0.00 \text{ m})}$ - 1.2 in (0.03 m)

High Pitch Limit:

30° ± 1°

POWERPLANT INSTRUMENT MARKINGS

Instruments	Red Radial Lower Limit	Yellow Arc Caution Range	Green Arc Normal Oper. Range	Red Radial Upper Limit
Tachometer (RPM)			1800 - 2400	2400
Manifold Pressure (inHg)		25 - 29.4	15 - 25	29.4
Oil Tempera- ture Indicator (° C)		93 - 118 below 74	74 - 93	118
Cylinder-Head Temperature Indicator (°C)		224 - 260	130 - 224	260
Oil Pressure Indicator (bar)	1.7	1.7 - 3.8 6.6 - 7.9	3.8 - 6.6	7.9
Fuel Gauges	0 (Zero)			
Fuel Pressure (bar)	0.83	0.83 - 0.96	0.96 - 3.1	3.1

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MAXIMUM CERTIFIED WEIGHT

Maximum Take-Off and Landing Weight Maximum Load in Baggage Compartment

2977 lbs (1350 kg) 110 lbs (50 kg)

CENTER OF GRAVITY LIMITS

Forward Limit:

89.3 in (2.2685 m) aft of datum (25 % MAC) at 2977 lbs (1350 kg)

86.5 in (2.1976 m) aft of datum (20 % MAC) at 2756 lbs (1250 kg)

86.5 in (2.1976 m) aft of datum (20 % MAC) at 2119 lbs (961 kg)

Aft Limit:

93.2 in (2.3677 m) aft of datum (32 % MAC) at 2977 lbs (1350 kg) 93.2 in (2.3677 m) aft of datum (32 % MAC) at 2119 lbs (961 kg)

Position of Reference Datum:

Fuselage Station 0.00 in (0.00 m)

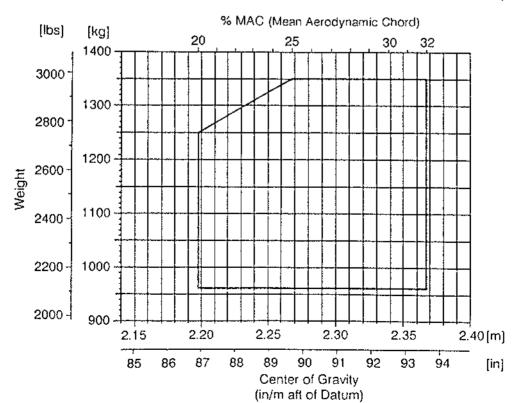
Mean Aerodynamic Wing Chord (MAC):

55.8 in (1.418 m)

Position of MAC:

between Stations 75.4 in (1.914 m)

and 131.2 in (3.332 m)



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MANEUVER LIMITS

This airplane is certificated in the "Normal" category. This category applies to aircraft, designed for normal operation (i.e. nonacrobatic). This includes any maneuver incident to normal flying, slow stall, lazy eights, chandelles and steep turns in which the angle of bank is not more than 60°. Acrobatic maneuvers including spins are prohibited.

FLIGHT LOAD FACTOR LIMITS

 $+3.8 \text{ g to } -1.52 \text{ g}^*$ Flaps retracted Flaps extended (15° und 30°) +2.0 g to 0.00 g

KINDS OF OPERATIONS LIMITS

The following types of operations are approved when the required equipment is installed and operational as defined in the KINDS OF OPERATIONS EQUIPMENT LIST.

- 1. VFR Day
- 2. VFR Night
- 3. IFR Day
- 4. IFR Night

Flights in known icing conditions are prohibited.

^{*)} Intended maneuvers with negative load factors are prohibited.

KINDS OF OPERATIONS EQUIPMENT LIST

This airplane may be operated in day or night VFR, or day or night IFR, when the appropriate equipment is installed and operable.

The following equipment list identifies the systems and equipment upon which type certification for each kind of operation was predicated. The systems and items of equipment listed must be installed and operable for the particular kind of operation indicated unless:

1. The airplane is approved to be operated in accordance with a current Minimum Equipment List (MEL) issued by the FAA.

or;

 An alternate procedure is provided in the Pilot's Operating Handbook and FAA approved Flight Manual for the inoperative state of the listed equipment and all limitations are complied with.

Numbers on the Kinds of Operations Equipment List refer to quantities required to be operative for a specified condition.

Note

The following systems and equipment list may not include all equipment required by the applicable operating requirements. It also does not include components obviously required for the airplane to be airworthy such as wings, empennage, engine, etc.

				·····	
OVOTER	VF	R Da			
SYSTEM and/or		VF	R Nig		
and/or COMPONENT			IFF	Day	
OOM! ONE!!					R Night Remarks and/
					or Exceptions
ELECTRICAL POWER					-
Battery DC Generator	1 1	1 1	1 1	1 1	
DC Generator Annunciator	1	1			
DC Load Meter	1	1	1	1	
DC Volt Meter	1	1	1	1	A COLOR
ENGINE INDICATIONS					
Cylinder Head Temperature					
Indicator Exhaust Cas Tomporature	1	1	1	1	At the state of th
Exhaust Gas Temperature Indicator	1	1	1	1	
Manifold Pressure Indicator	1 1	1			
RPM Indicator	1	1	1	1	
ENGINE OIL					\$
Oil Pressure Annunciator	1	1	1	1	
Oil Pressure Indicator	1	1	1	1	
Oil Temperature Indicator	1	1	1	1	
FLIGHT CONTROLS					
Flap Position Indicator	1	1	1	1	
Flap System Elevator Trim Tab	1	1	1	1	
Position Indicator	1	1	1	1	
Stall Warning System	1		1	1	

	VFF	R Day	,		:
SYSTEM	VFR Night				
and/or		IFR Day			
COMPONENT			<u> </u>	IFR	Night Night
		<u> </u>			Remarks and/
			ļ		or Exceptions
ELIQUE INCEDIMENTO					
FLIGHT INSTRUMENTS	1	,	,		
Airspeed Indicator Altimeter		1 1	0	1 0	T A STATE OF THE S
Magnetic Compass		1	1	1	
Outside Air Temperature	'	'	'	'	
Indicator	0	0	1	1	
Sensitive Altimeter	ŏ	Ŏ	1		
Vertical Speed Indicator	1	1	1	1	
FUEL					
Auxillary Fuel Pump System	1	1	1	1	
Fuel Flow Indicator	1 1				
Fuel Pressure Indicator			1		
Fuel Quantity Indicator	2	2	2	2	
Fuel Tank Selector Valve	1	1	1	1	
ICE AND RAIN PROTECTION					
Alternate Static Air System	0	lo	1	1	
Pitot Heat	Ŏ	Ŏ	1	1	
LANDING GEAR					
Emergency Extension Valve	1	1	1	1	
Landing Gear Hydraulic	'		'	, '	
Power Pack	1	1	1	1	
Landing Gear Position					
Indicator Lights	4	4	4	4	
Landing Gear Warning Horn	1	1	1	1	

	VF	R Da	y		
SYSTEM VFR Night					
and/or		IFR Day			,
COMPONENT				IFR	? Night
					Remarks and/ or Exceptions
LIGHTS Cockpit and Instrument					
Lighting System	0	1	0	1	<u> </u>
Landing Light	1	1	1	i	
Position Lights	0	3	0	3	
Anticollision Lights					
(incl. Strobes)	0	3	0	3	
NAVIGATION INSTRUMENTS					
Gyroscopic Rate of Turn Indi-			,		
cator with Slip-Skid Indicator Gyroscopic Bank & Pitch	0	0	1	1]
Indicator	0	0	1	1	
Gyroscopic Direction Indicator	"	`	'	'	
(or equivalent)	0	0	1	1	
PROPELLER	<u> </u>				
Propeller Governer	1	1	1	1	
Propeller Spinner	1	1	1	1	
PUBLICATIONS		!		ĺ	
LBA Approved Airplane					
Flight Manual	1	1	1	1	
g	'	,	'	'	
RESTRAINT SYSTEM					
Seat Belt (per Seat)	1	1	1	1	
Shoulder Harness (per Seat)	1	1	1	1	
VACUUM SYSTEM			į		
Instrument Air System		_	4		
Suction Gauge	0	0	1	1	
	<u> </u>				

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MAXIMUM FUEL CAPACITY

	Left Tank	Right Tank
Total Capacity	1251 (33.0 US Gal)	1251 (33.0 US Gal)
Unusable Fuel	71 (1.8 US Gal)	71 (1.8 US Gal)
Usable Fuel	1181 (31.2 US Gal)	1181 (31.2 US Gal)

Maximum permissible difference between

left and right tank

501 (13.2 US Gal)

EXHAUST GAS TEMPERATURE LIMITS

Leaning of the fuel mixture is not allowed at continuous power of more than 85%.

FLIGHT CREW

Minimum	Flight Crew:	1
	Number of Usable Seats:	

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OTHER OPERATIONAL LIMITATIONS

Permissible Wing Flap Settings:

Take-Off Range: 0° or 15°

PLACARDS

The placards and markings installed in this alrolane contain operating limitations which must be complied with when operating this airplane in normal category. Other limitations which must be complied with when operating this airplane in this category are contained in the airplane flight Manual.

No acrobatic maneuvers, including spins,

Operation under known icing conditions is prohibited.

This alteraft is approved for day/night VFR and IFR flights if the required equipment is installed.

> Right Side of Panel (bottom left)

MAXIMUM SPEED KIAS ALTITUDE 193 MSL-12000ft 16000ft 20000ft

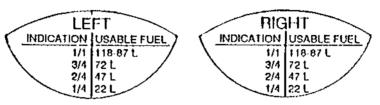
Left Side of Panel (above ignition switch)

MANEUVERING SPEED MAXIMUM SPEED FOR LAN-**DING GEAR OPERATION**

Left Side of Panel (below ignition switch)

NO INDICATION OF USABLE FUEL QUANTITIES ABOVE 88 L FOR EACH TANK

Left Side of Panel (above fuel indicators)



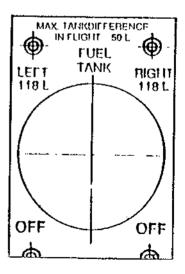
Left Side of Panel (part of fuel indicators)

NO SAFE USE OF FUEL REMAINING, WHEN INDICATOR REEDS ZERO IN LEVEL FLIGHT

Left Side of Panel (below fuel indicators)

TANK FILLER CAP
PULL TO OPEN
DO NOT OPEN DURING FLIGHT I

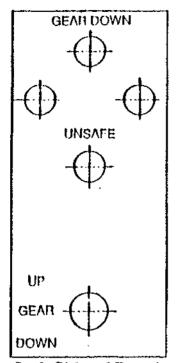
Entrance Hatch (beside the front seats)



Central Pedestal (center)

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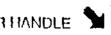
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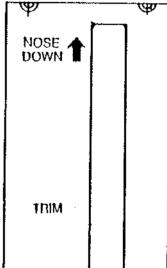
EMERGENCY GEAR EXTENSION 1. PULL GEAR ACT CINCUIT BREAKER

2. GEAR SWITCH DOWN

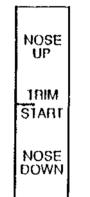
3. PULL UP EMERGENCY GEAR HANDLE



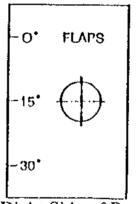
Central Pedestal (pilot's side)



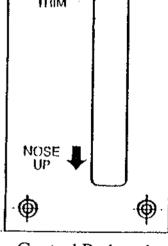
Left Side of Panel (beside the avonics stack)



Right Side of Panel

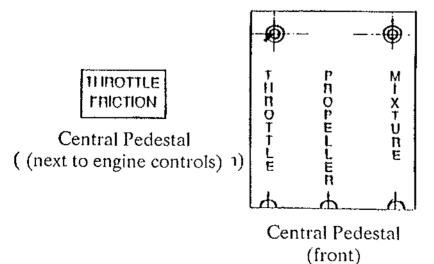


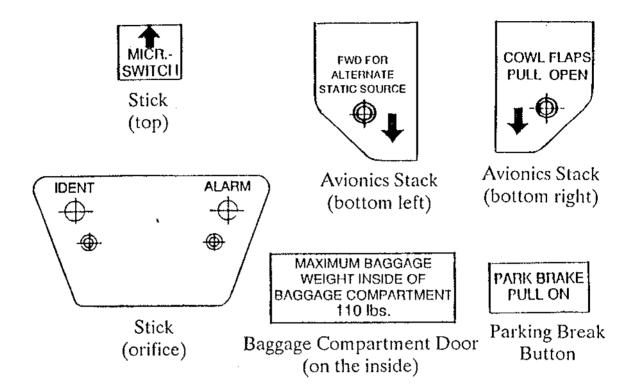
Right Side of Panel



Central Pedestal (rear)

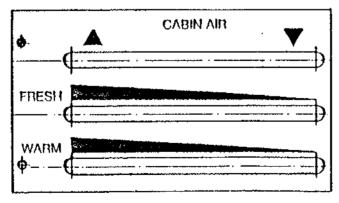
(next to trim indication) (next to flap position indication)



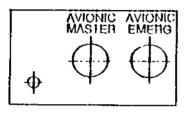




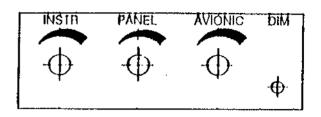
Entrance Hatch (pilot's side)



Right Side of Panel (bottom)



Avionics Stack (above avionics)



Avionics Stack (above avionics)